

Full-Scale Crash Test of a Civil Helicopter

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Contents

1. Overview of crash test

2. Test method

- Article
- Facility
- Test conditions
- Measurement method

3. Test results

4. Conclusion

Overview of Crash Test

- Second crash test of a helicopter in Japan
- Crash trajectory determined by guided rail method
- Cooperation with Mitsubishi Heavy Industries, Ltd. (MHI)
- Test on February 25, 2004 at Aerospace Center Aerodrome Branch of JAXA



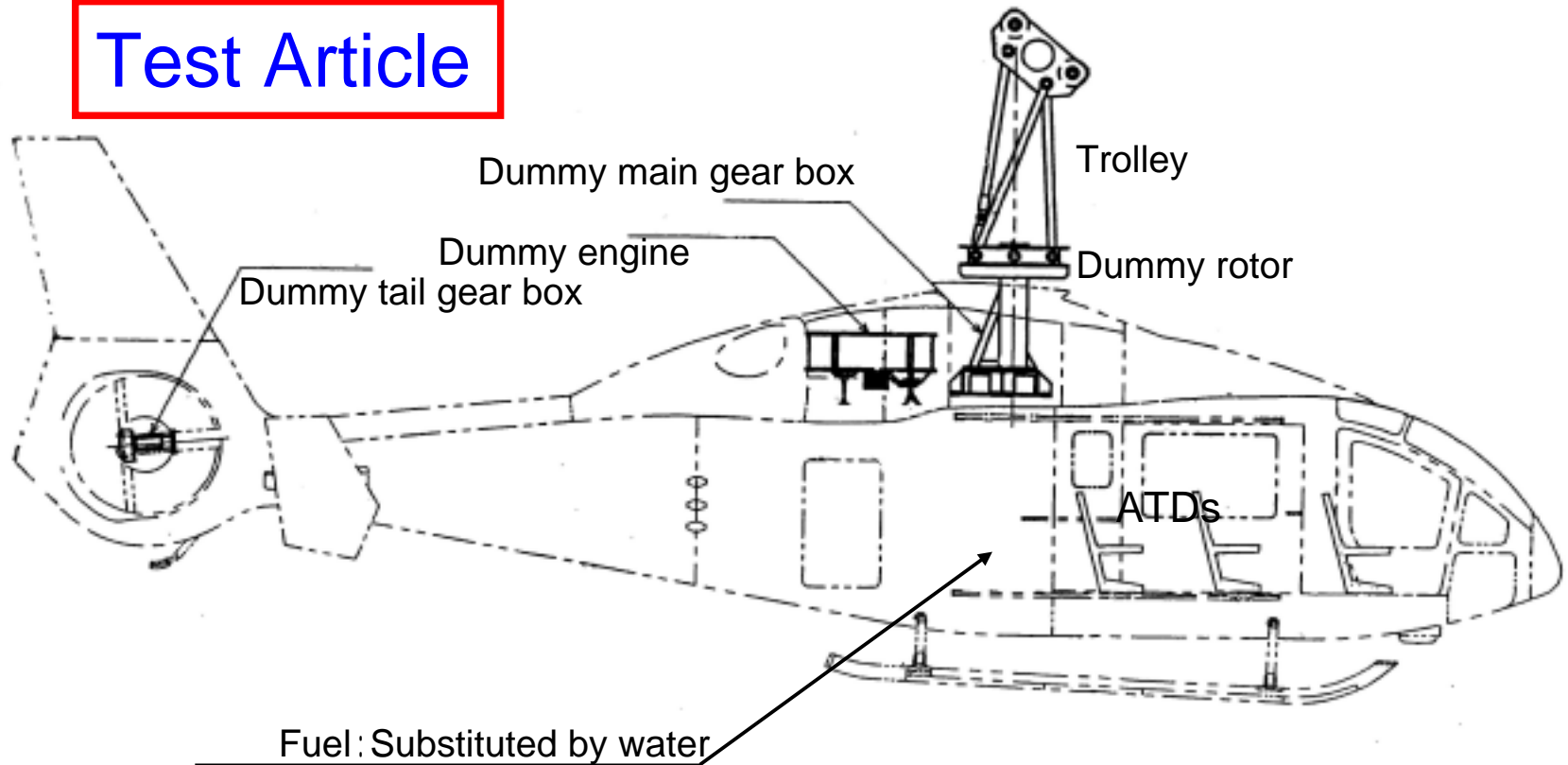
Overview Video Picture of the Test



Objective

1. Determine the impact response of a helicopter to a full scale crash test
2. Validate helicopter analytical computer model against actual crash test data
3. Develop guided rail method to conduct crash testing
4. Evaluate performance of NAL developed pilot seat shock-absorbing device

Test Article



A prototype of Mitsubishi MH2000A multi-purpose helicopter

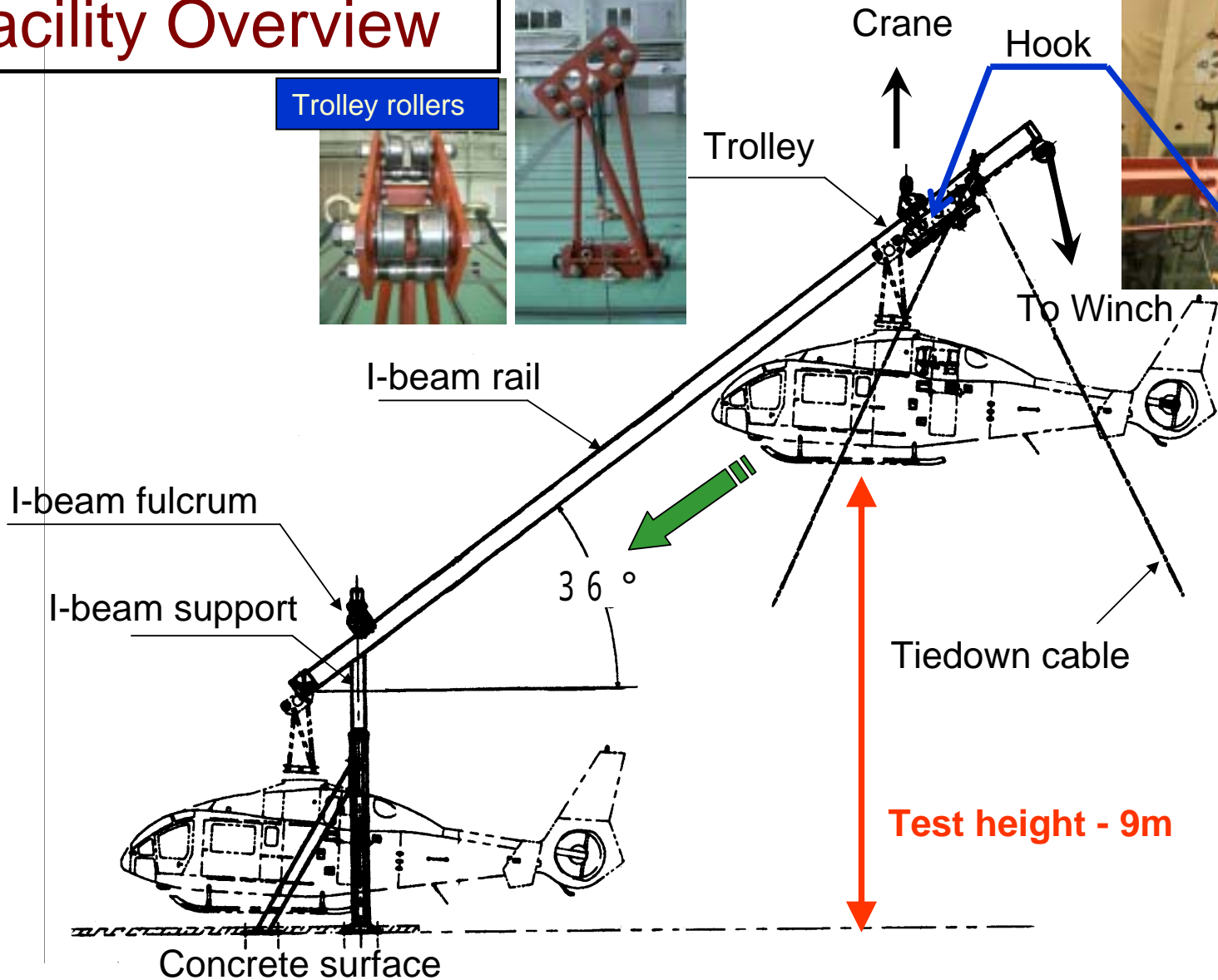
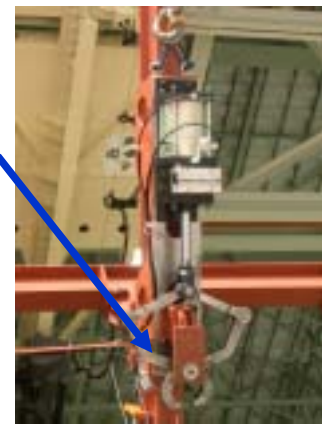
Total weight : 4,500 kg (10,000 lbs)

Number of seats : (pilots 2, passengers 8)

Anthropomorphic Test Dummies
(7 Hybrid II ATDs, 1 Hybrid III ATD)

Facility Overview

Trolley rollers



Suspended Test Article



Test Conditions

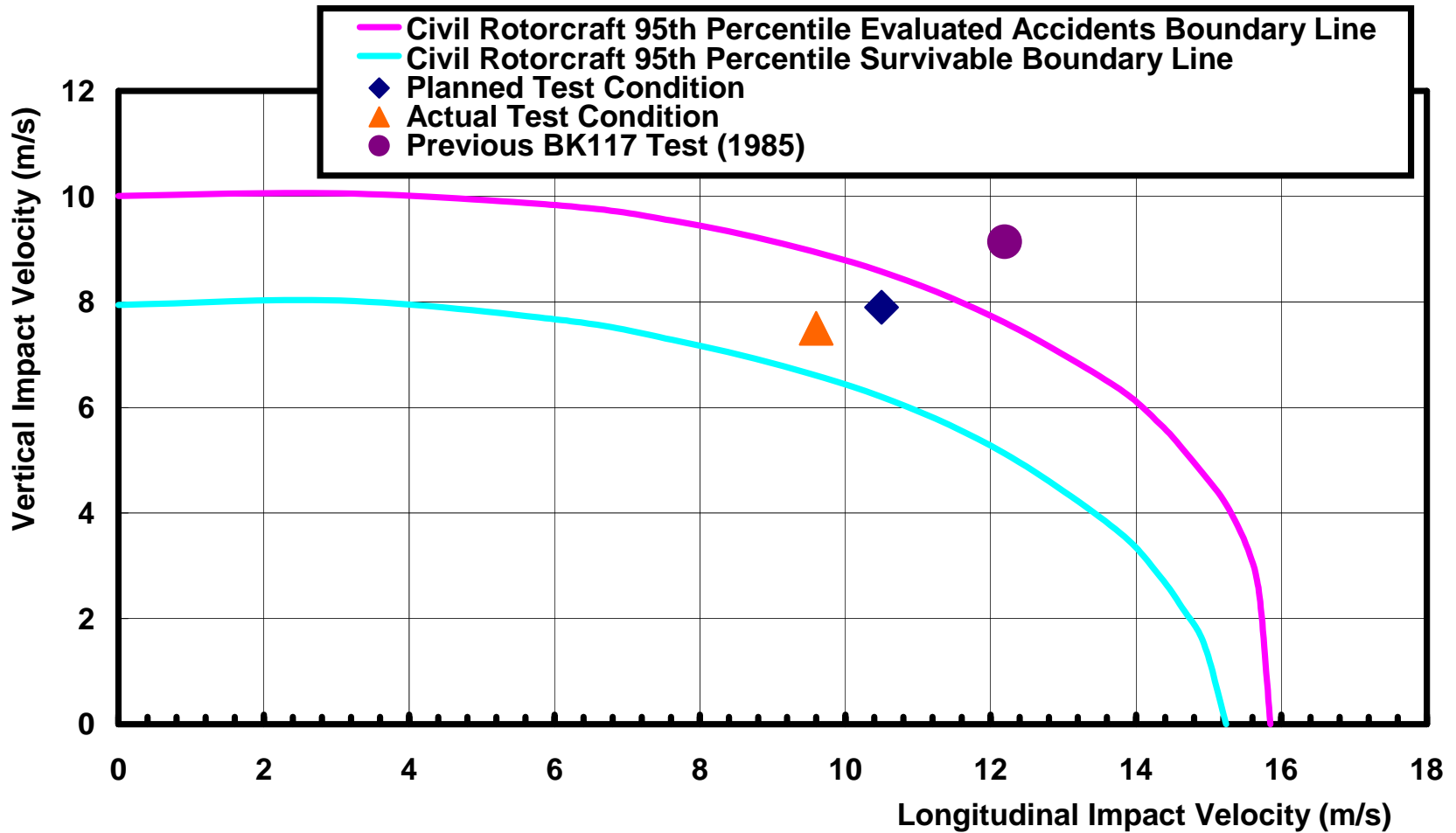


Figure: Comparison of Impact Velocity Envelope

Test Conditions (Continued)

Velocity components (Planned)

Vertical : 7.9 m/s (26 ft/sec)

Longitudinal : 10.5 m/s (34 ft/sec)

(derived from the vertical component of the dynamic seat test condition and an approach angle of 37°)

Pitch angle

+4 °

(Half of 8° angle which a line from the skid to the tail makes with the horizontal)

Test article slides down I-beam (14.8 m), drops on concrete surface

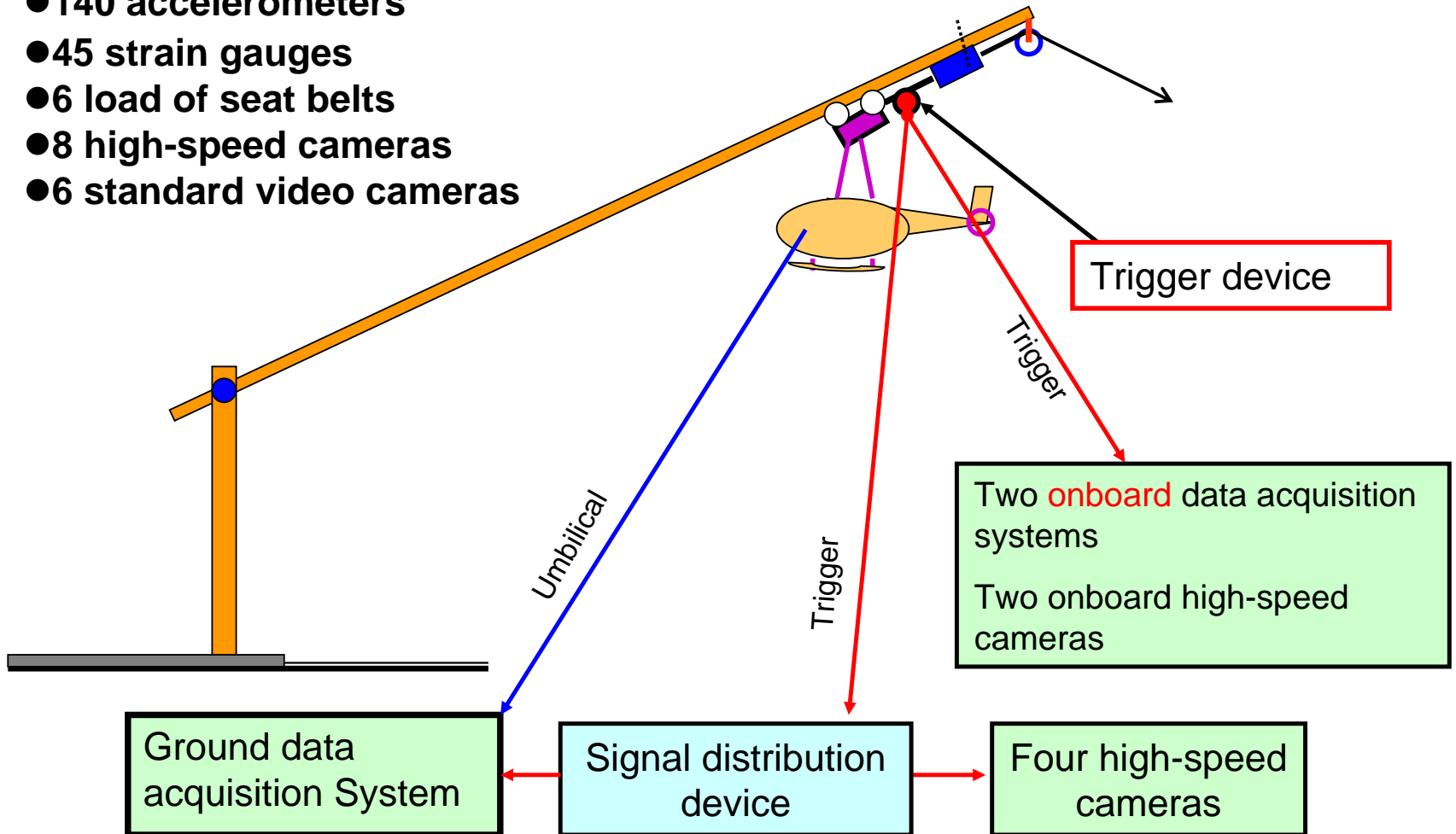
Beam angle : 36°

Height of test : 9m

Height of free fall : 0.3m (from end of beam)

Data Acquisition

- 140 accelerometers
- 45 strain gauges
- 6 load of seat belts
- 8 high-speed cameras
- 6 standard video cameras

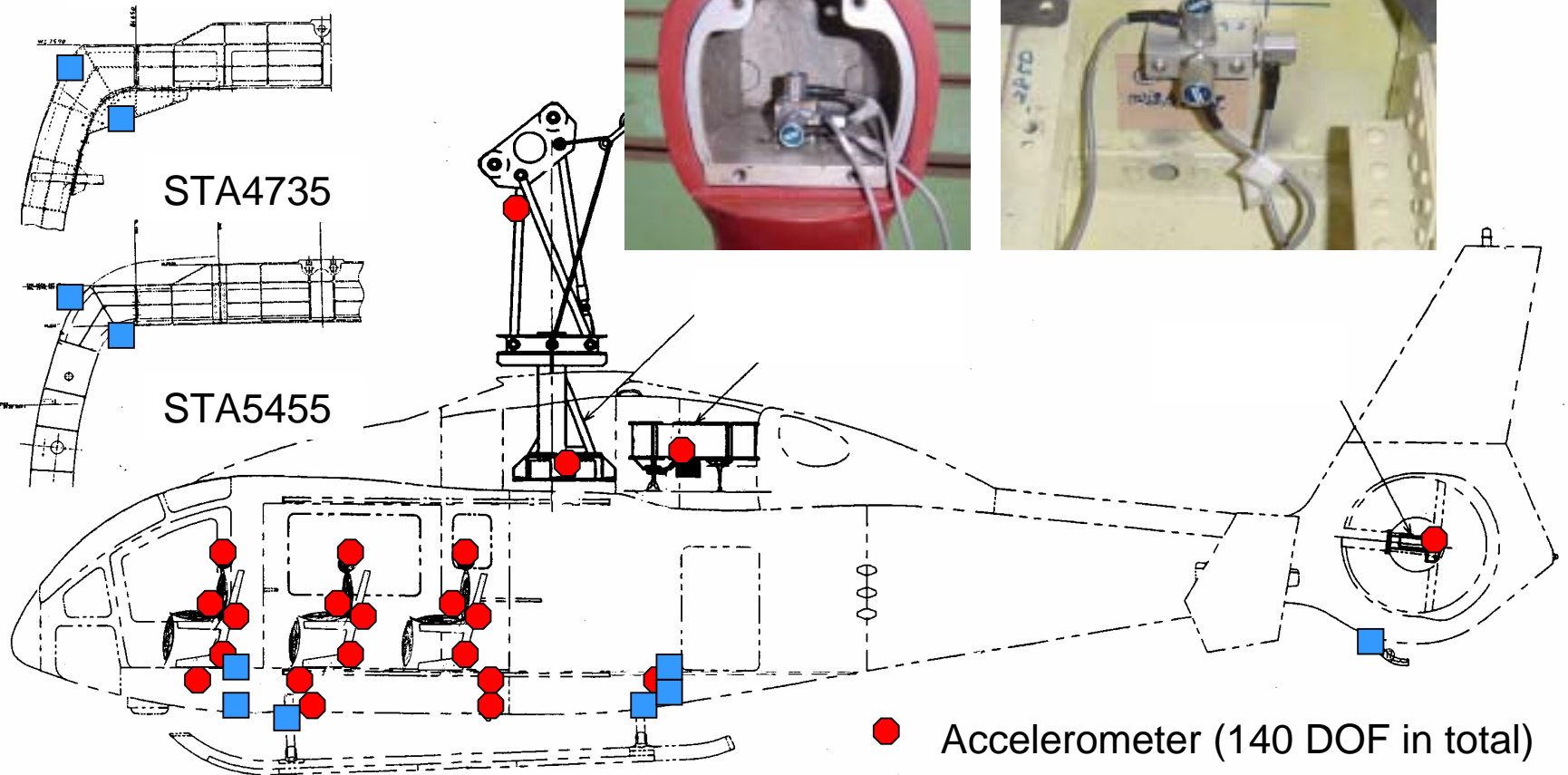


Sensor Installation

Main Frame

STA4735

STA5455



- Accelerometer (140 DOF in total)
- Strain Gauge (45 DOF in total)

Pretest Overview of Onboard Measurement Devices

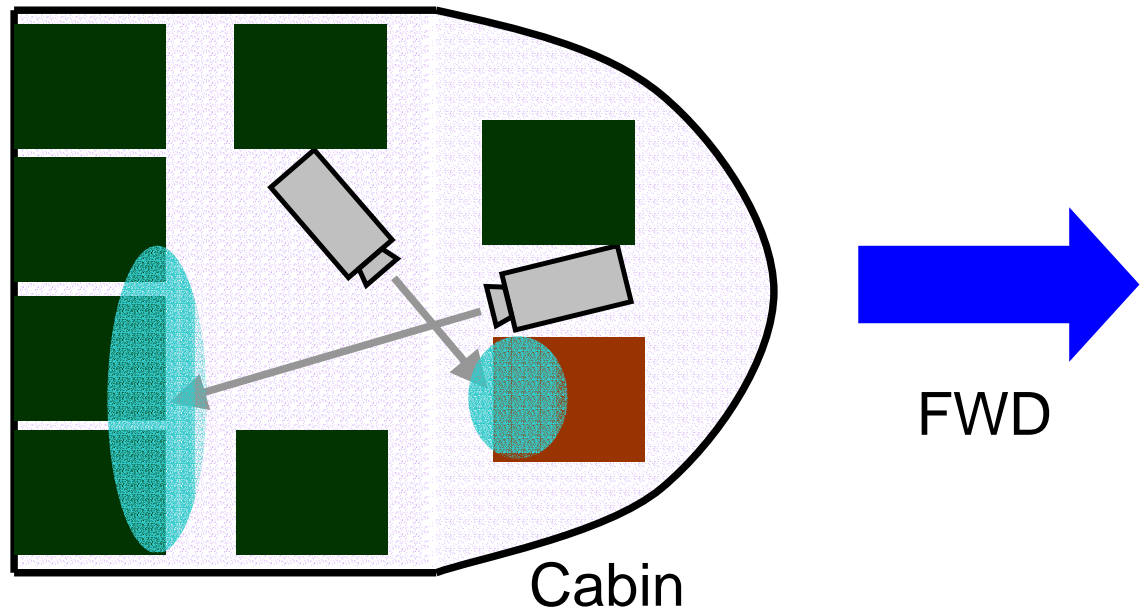


Onboard data acquisition systems



Onboard high-speed camera

On-board High Speed Camera Position



Standard MH2000 seat

Pilot seat with JAXA
energy absorbing device

High speed camera

Four Meter High Mock-up Test



Mock-up test:

2m

4m

5m Step-up verification of our test method¹⁵

Video Picture of the Test from Backside



High-Speed Video Picture of the Test



High-Speed Video Picture of the Test



Offered by MHI

Overall Post-test



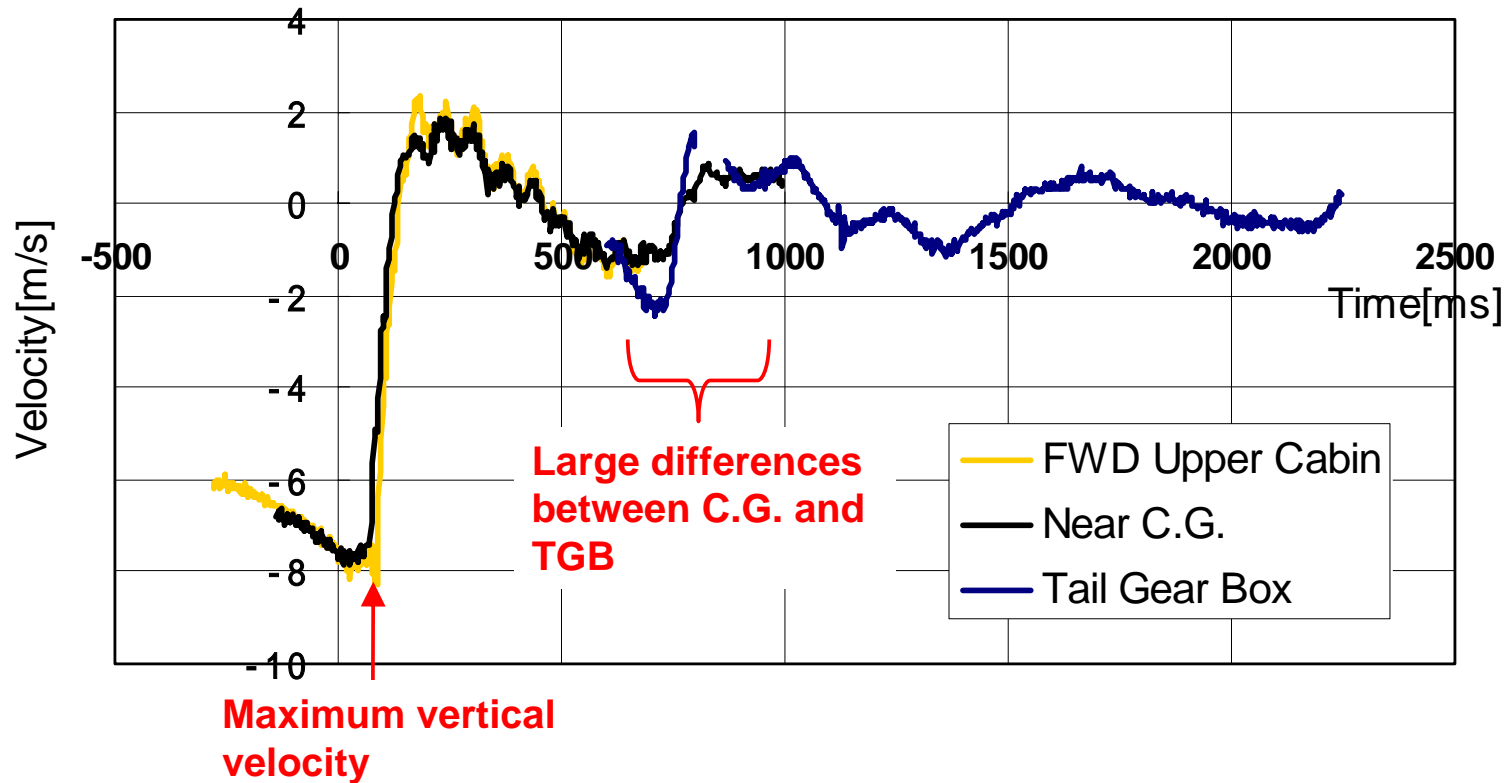
Test Results from Analysis of High-Speed Camera Pictures

- Impact velocity components

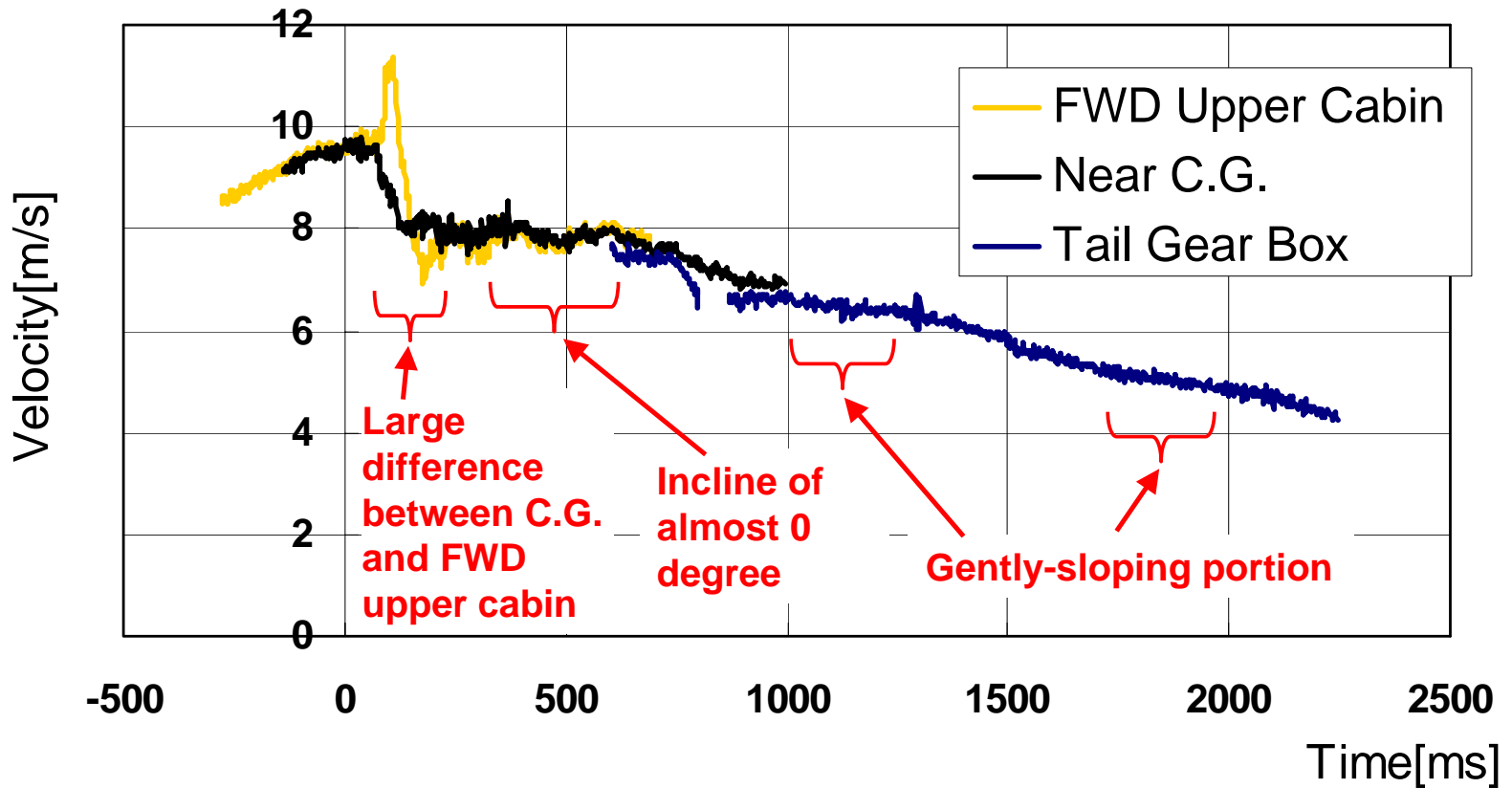
Vertical: Planned 7.9m/s	Actual 7.5m/s (- 5.1% error)
Horizontal: Planned 10.5m/s	Actual 9.6m/s (- 8.6% error)

- Pitch angle: Planned +4 ° Actual +2.7 °

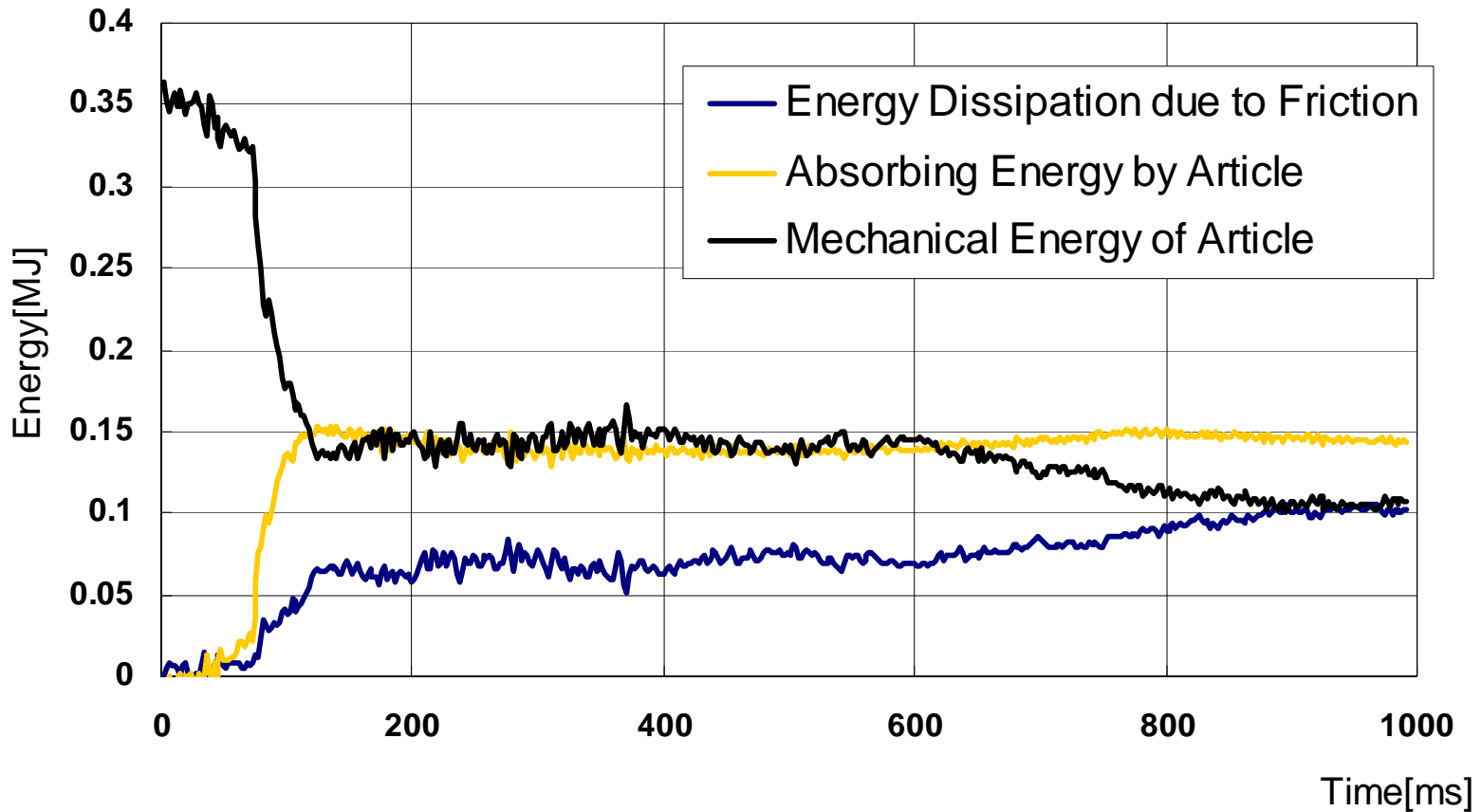
Vertical Velocity



Horizontal Velocity

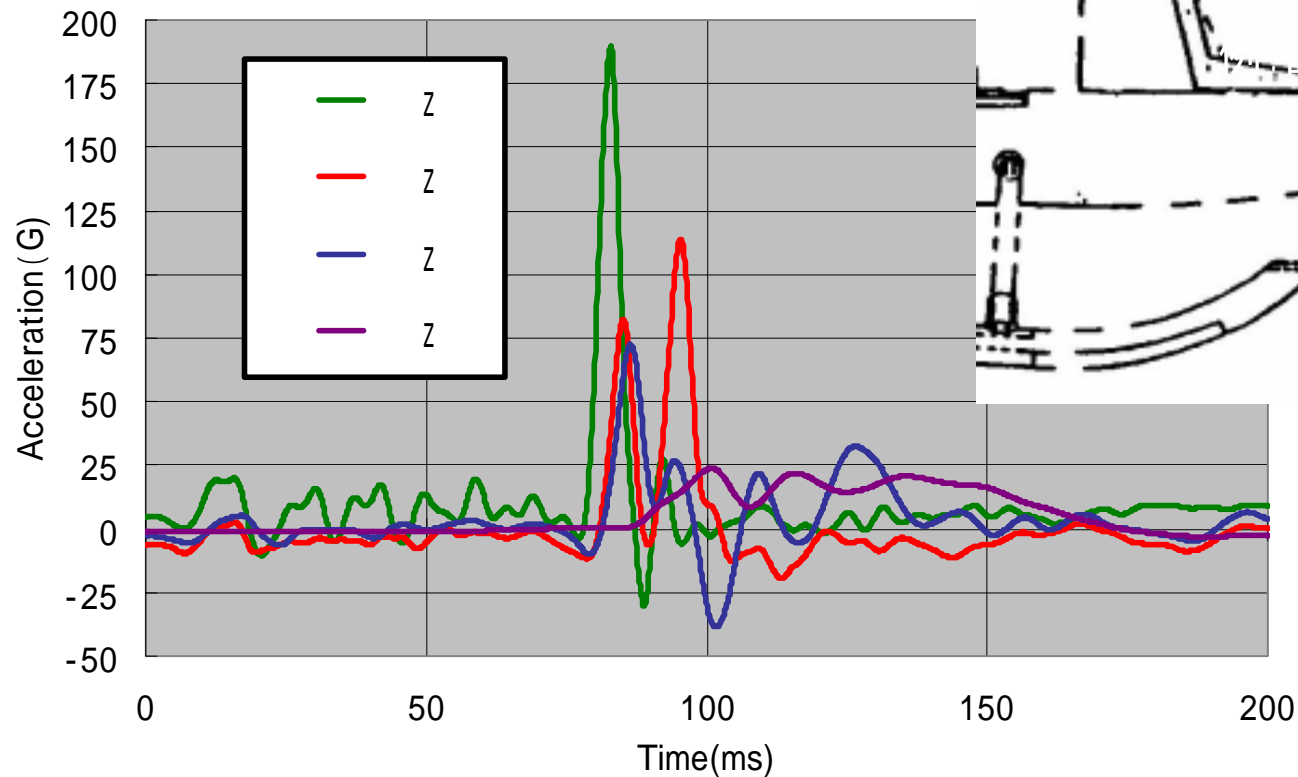
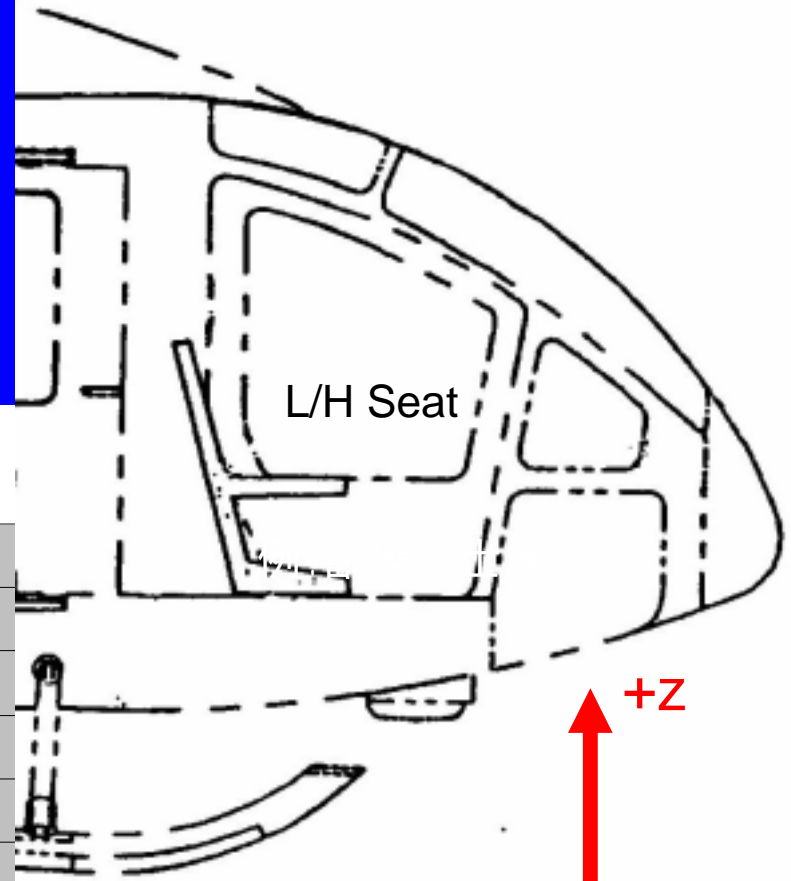


Energy Dissipation



Sample Accelerations

T_0 = landing skid impact.



Standard MH2000 Seat



Wire bender energy absorber



JAXA Energy Absorbing Device



Pretest



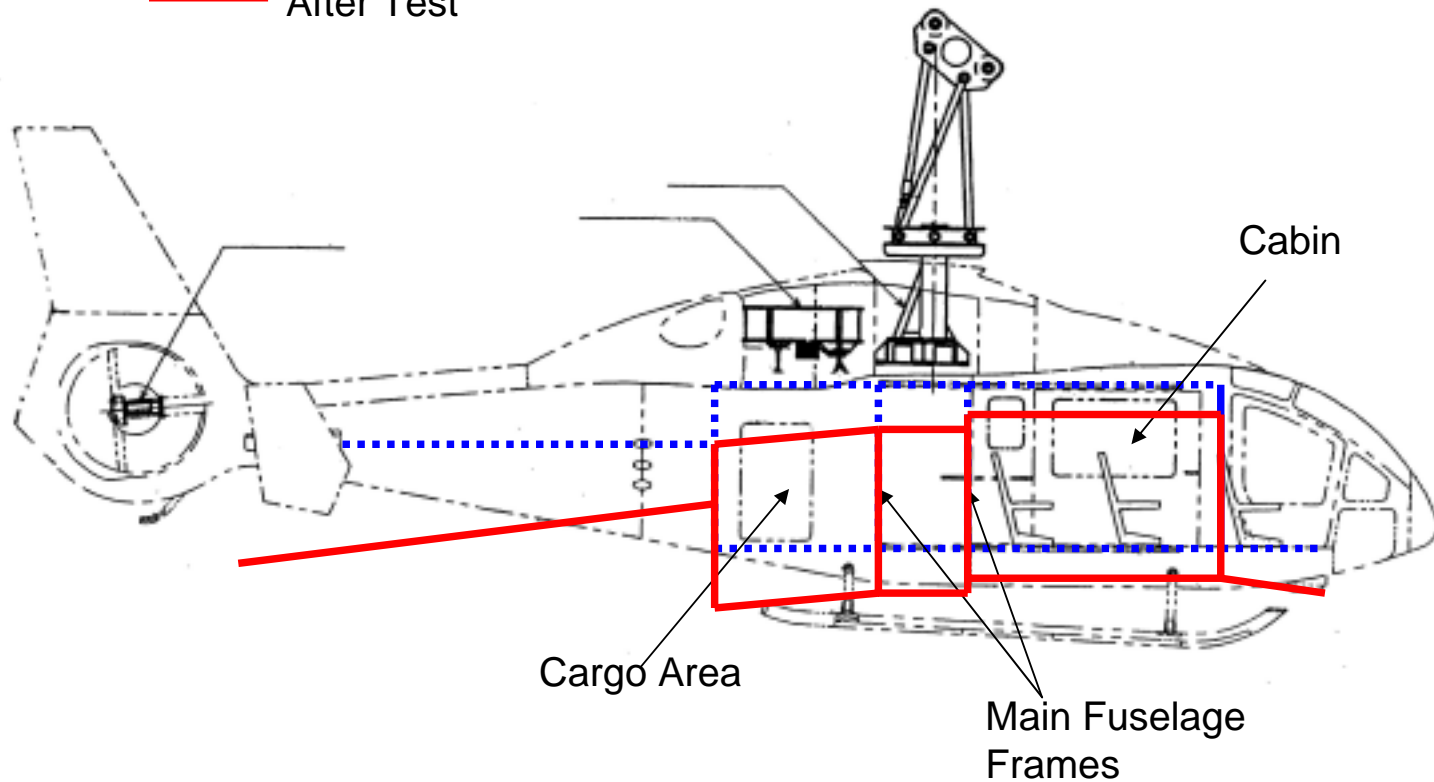
Post-test

Did not function properly due to off-axis loading

Deformation of Structure after Impact

..... Before Test

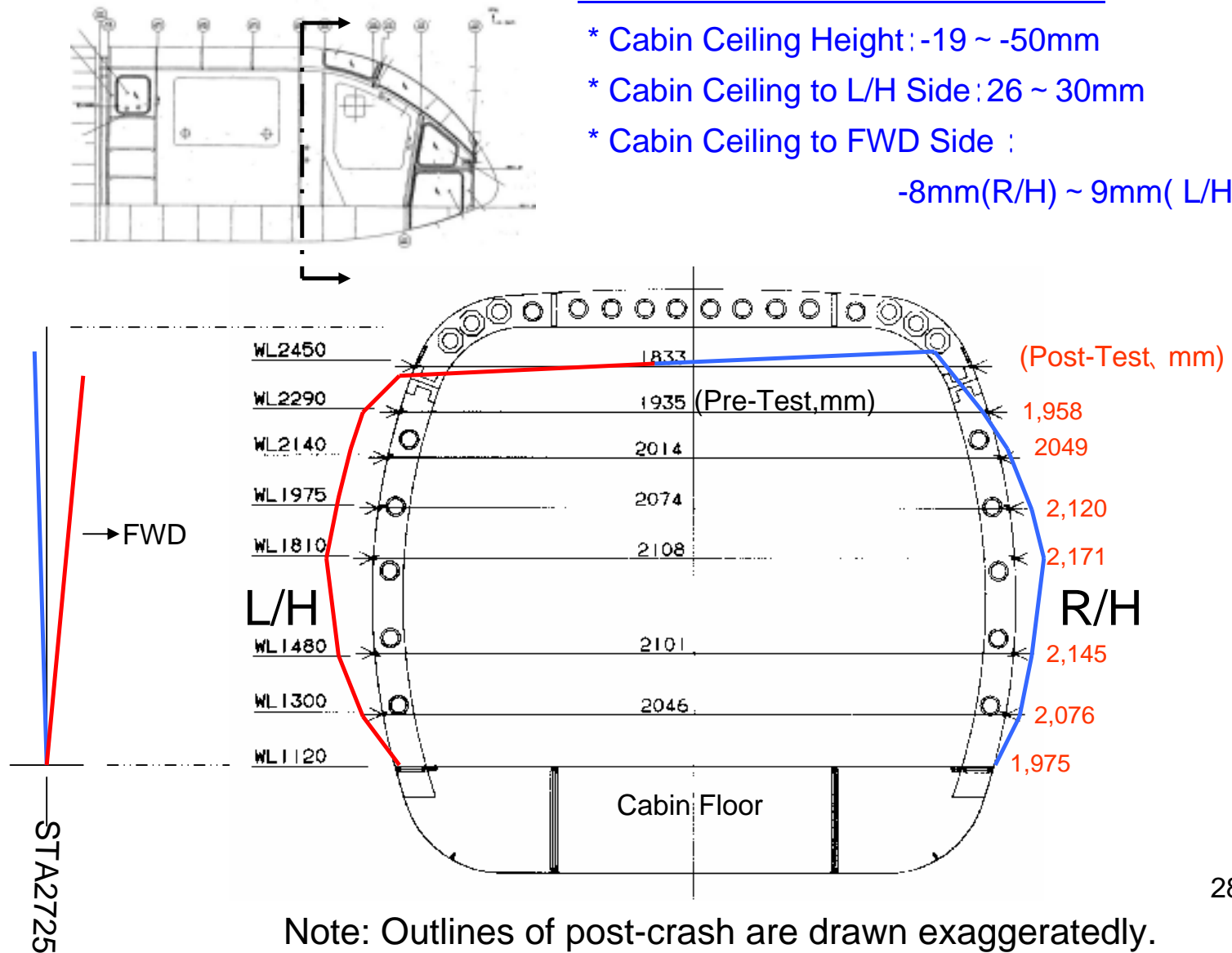
— After Test



Cabin Deformation of STA 2725 Cross Section

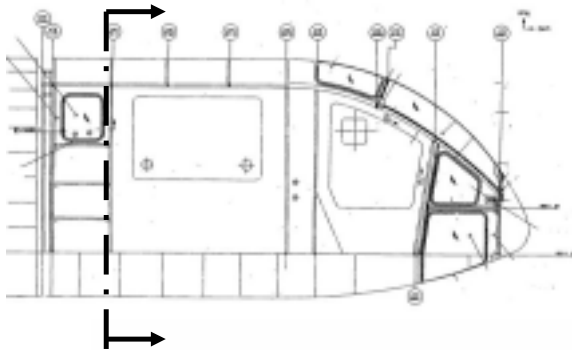
Crash Distances of the Cabin

- * Cabin Ceiling Height: -19 ~ -50mm
- * Cabin Ceiling to L/H Side: 26 ~ 30mm
- * Cabin Ceiling to FWD Side :
-8mm(R/H) ~ 9mm(L/H)mm



Note: Outlines of post-crash are drawn exaggeratedly.

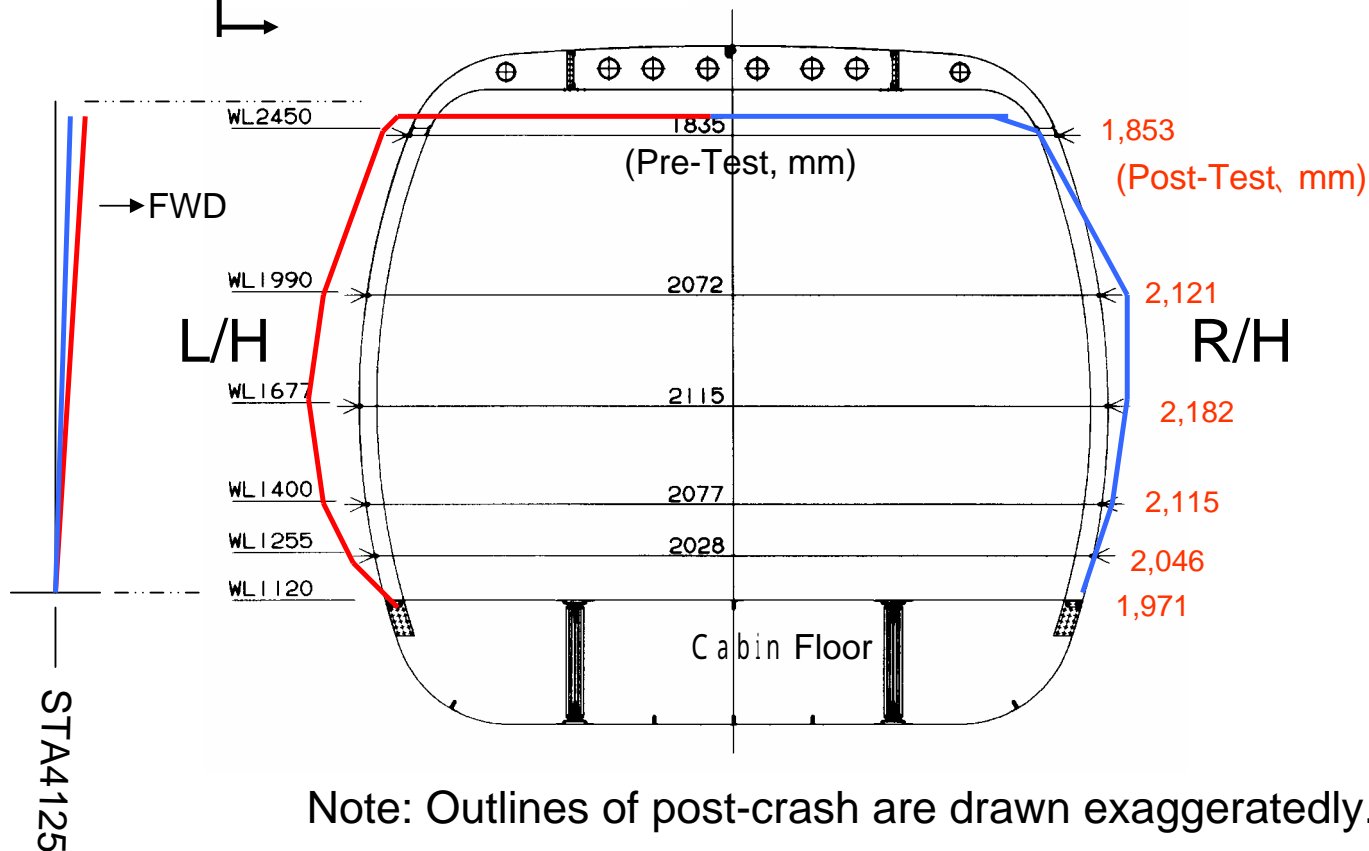
Cabin Deformation of STA 4125 Cross Section



Crash Distances of the Cabin

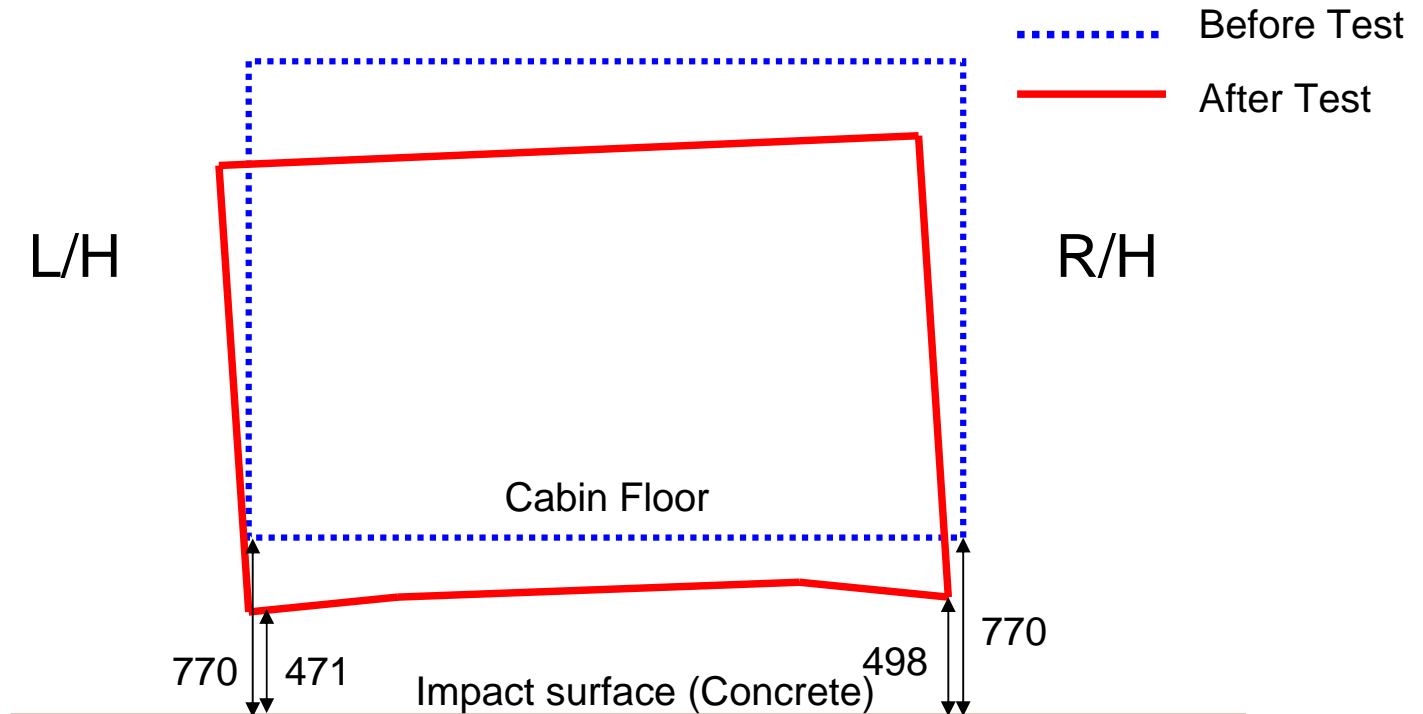
- * Cabin Ceiling Height: -22 ~ -23mm
- * Cabin Ceiling to L/H Side: 23 ~ 24mm
- * Cabin Ceiling to FWD Side :

8mm(R/H) ~ 13mm(L/H)mm

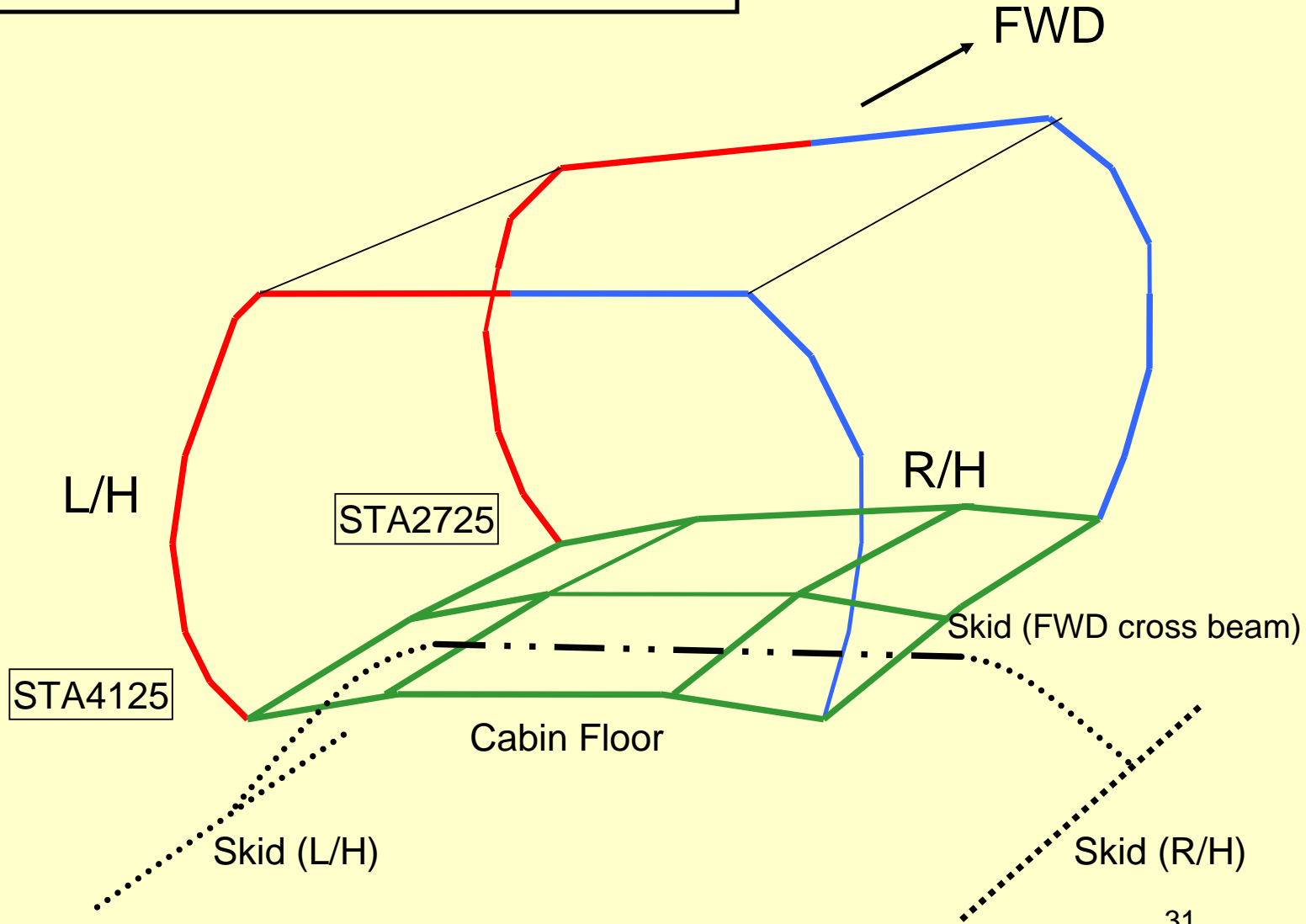


Schematics of Rear View of Cabin

Cross Section Near STA 4010



Overview of Cabin Deformation



Conclusion

- Able to successfully acquire real time helicopter crash data for verifying full-scale helicopter analytical computer model and for being utilized as reference for new helicopter design
- Able to utilize guided rail method to conduct a crash test